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RESEARCH MEMORANDUM

AGE AND USAGE FACTORS AFFECTING MISSILE RELIABILITY: A STUDY BASED ON NIKE-AJAX EXPERIENCE (U)

> William H. McGlothlin David S. Stoller

> > RM-2560

April 15, 1960

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SUMMARY

This study examines the relationship between environmental history and missile firing success, based on a sample of 2585 NIKE-AJAX firings. The MIKE-AJAX was chosen because the data comprised the only available sample on U.S. missiles large enough to permit the kind of analysis done in this study. The primary data came from the historical logbooks which accompany each missile throughout its life. The major variables were: (1) date of manufacture and rebuilding (if any) for both the missile and the missile guidance unit, (2) time missile was on a NIKE battery site, (3) time missile guidance unit was operated both before and after rebuild, and (4) outcome of firing. Supplemental data were gotten from the NIKE firing summaries maintained by the Douglas Aircraft Company. They covered (1) malfunctions listed by system, and (2) the type of crew conducting the firing.

The split-sample technique was used in the analysis, with the 2585 firings divided into two approximately equal samples by odd and even serial numbers. The first half-sample was used for formulating hypotheses about relationships between firing reliability and the other variables. The second half-sample was used to test the statistical significance of these hypothetical relations.

The results of the study were that:

- (1) Even after field-level overhaul of the missiles immediately before firing, missile usage on tactical sites, particularly as measured by the guidance-unit operating-time, lowers firing reliability.
- (2) After a field-level overhaul, missile age factors have a negligible effect on firing reliability, if any; particularly, there appears to be no post-overhaul effect of prior time in depot storage.

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(3) Except for missile-guidance malfunctions, the reliability of rebuilt missiles compares favorably with the reliability of those not rebuilt.

These results should be interpreted in the light of maintenance policies which provide a field-level maintenance inspection and repair—as—necessary immediately before the firings. It is probable that missile age and usage have greater unfavorable effects on combat reliability than those assessed in this study; but the field-level overhaul intervening between removal from the site and firing allows us to measure only the residual effects subsequent to this maintenance action. Test firings are normally conducted only after very thorough checkout and maintenance procedures. Tactical missiles, of course, would not have the benefit of this sort of maintenance immediately before firing.

This points up a problem that is common to most missile programs. The use of test firing recults can lead to unrealistic estimates of combat capability, and can tend to obscure empirical measures of the effects of age and usage on firing reliability which, in turn, would be useful in determining preferred maintenance procedures for combat missiles. At the time a missile weapon system is considered operational, it would seem advicable to initiate test firings under conditions (1) which more closely simulate those of a combat environment, and (2) such that the effects of pre-launch environments on missile reliability can be more accurately assessed.

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ACKNOWLEDGMENTS

The authors wish to express their indebtedness to the Field Service Division of Redstone Arsenal who extracted, and made available to us, the major portion of the data used in this study. A. A. Stewart and Paul Kane were especially helpful in this connection. In addition, the Douglas Aircraft Company make available some useful supplemental data. W. P. Anderson, Sadako Hayase, and Claire Buccella capably handled the numerical analysis of the data.

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I. INTRODUCTION

One of the major problems in determining a preferred policy for ICBM and other missile operation and maintenance is the difficulty of estimating the effects of age and usage on firing reliability. There have been some efforts to measure these effects on failure rates at missile checkout, but the typically small number of firings has yielded very little information concerning the effect on firing reliability. The NIKE-AJAX program, however, did encompass a large number of firings and thus yielded a large sample of environmental histories and firing outcomes. This study uses 2585 of these firings to explore the relationship between pre-launch environments and missile firing success.

The ago effect examined here will be limited to "permanent" changes in missile firing reliability as a function of age, as opposed to "temporary" changes which may be detected and corrected by routine checkout procedures. Since all of the NIKE missiles received a thorough checkout immediately prior to firing, any detectable age effect would be a residual, or "permanent" affect.

The term "usage" in this paper will mean the maintaining of the missile in a ready or alert state, the operation of certain portions of the missile for checkout purposes, and its use for simulated combat exercises. The effect of usage on firing reliability is vitally important in determining missile

^{*}C. E. Parker, "Some Effects of the Logistics Environment of an Air-to-Air Missile," <u>Proceedings of the Symposium on Guided Missile Reliability</u>, Part I, December, 1958, Dept. of Defense, pp. 111-116. W. H. McGlothlin and P. R. Yorshis, <u>Measuring Missile Checkout Reliability Over Time</u> (U), The RAND Corporation, Research Memorandum RM-2243, August 15, 1958 (Secret).

P. J. Doyle and W. W. Szkil, Effects of Storage and Testing on Terrier Success Rates, U. S. Naval Ordnance Laboratory, Corona, Calif., May, 1956 (Confidential).

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maintenance policies. If operating the missile for checkout purposes tends to reduce firing reliability, then this loss must be balanced against gains in overall combat capability resulting from the prompt detection of temporary age effects, i.e., changes affecting reliability which occur while the missile is standing in a ready, but inert state. Similarly, if using the missile for crew training reduces reliability, then this loss should be evaluated against possible gains in combat effectiveness resulting from training personnel on tactical missiles.

Because of limitations in the kinds of data recorded in the histories, and the absence of experimental control in the samples, particularly with respect to maintenance policies (see p. 35), this study does not yield sufficiently complete planning information on the factors affecting missile combat capability to permit an assessment of the optimal use of the missile for checkout and training purposes. However, the measures of the effect of age and usage factors on firing reliability should prove generally useful in this area. This is particularly true in view of the unusually large sample studied here.

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[&]quot;See D. S. Stoller and R. L. Van Horn, Management Information for the Maintenance and Operation of the Strategic Missile Force, The RAND Corporation, Research Memorandum RM-2131, April 30, 1958.

II. DATA

The majority of the data were extracted from the NIKE-AJAX logbooks by the Field Service Division at Redstone Arsenal. A logbook accompanies each missile throughout its life and provides a history of the environment to which the missile is exposed from manufacture to firing. Supplemental data were obtained from the firing summaries maintained by the Douglas Aircraft Company. The following are the data variables:

I. Redstone logbooks

- 1. Missile manufacturing date
- 2. Missile guidance unit manufacturing date
- 3. Missile rebuild date (if any)
- 4. Missile guidance unit rebuild date (if any)
- Time on factical site before rebuild, or since new if no rebuild occurred
- 6. Time missile guidance unit was operated before rebuild (if any)
- Time missile guidance was operated after rebuild, or since manufacture if no rebuild occurred
- 8. Date of firing
- 9. Location of firing
- 10. Outcome of test.

II. Douglas data

- Missils and missile guidance unit serial numbers at the time of manufacture
- 2. Type of crew conducting firing
- 3. Malfunctioning system when known.

The completeness of the logbook entries for the above variables ranged from 100 per cent for date of manufacture and cutcome of test, to approximately 80 per cent for time-on-site and time the missile guidance unit was operated. We had no means of assessing the accuracy of the entries.

Table 1 gives the total population of NIKE-AJAX firings for the period from January, 1953 to December, 1958, the number included in the Redstone sample, and the number for which both Redstone and Douglas data were available.

^{*}The missile guidance unit is contained in the missile and should not be confused with the ground guidance radars and computer.

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Total and sample nike-ajax firings: January, 1953 - December, 1958

Year	Totala	Redstone Sample	Both Redstone an Douglas Samples				
1953	79	٥	a				
1954	297	20	2				
1955	568		367				
1956	711	374 600	592				
1957	794	771	657				
1958	874	713	391				
Data Missing	-	107	104				
Total	3323	2585	2113				

^{**}The total population of firings was obtained from the quarterly Project NIKE Firing Tests Summary, Bell Telephone Laboratories, December, 1954 to Pourth Quarter, 1958 (Confidential).

Both the Redstone and the Douglas data include firing outcomes. Redstone scored 74 per cent of the total firings as successful, while Douglas listed 68 per cent successful. The following table shows the extent of agreement between the Redstone and Douglas scorings:

SCORING COMPARISON

Douglas	Radatone Scoring								
Scoring	Successful	Unsuccessful	Total						
Successful	1407	39	1446						
Unsuccessful	140	527	667						
Data Kissing	353	119	472						
Total	1900	685	2585						



The Redstone data were given preference when the two scorings were discrepant, primarily so that we might use the total sample of 2585 firings rather than the 2113 for which both Redstone and Douglas data were available.*

The Douglas data identified malfunctioning systems as shown in the following table. The Redstone data did not include system malfunctions in firings which were scored successful.

System

Armament	55
Booster	
Flight Coverage	1
Ground Guidance	
Missile Guidance Unit	127
Kydraulic	
Launcher	
Propulation	82
Missile (general)a	<u> 297</u>
Total ^b	638

These malfunctions are attributed to the missile rather than the ground equipment, but the specific system is unknown.

bIncludes multiple system malfunctions in a single firing. All the malfunctions listed here were considered to have contributed to the failure of the mission.

Since this study is primarily concorned with relating missile history to the firing outcome, the question arises as to whether the sample should include firings which were scored unsuccessful because of ground-system malfunctions. It was decided to retain these firings because (1) there may

^{*}The Redstone critarion of a successful test was "a missile released within launching time (on or before the scheduled launch time) that bursts within 225 feet of the target." The Douglas criterion was more stringent.



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have been an interaction between the performance of the missile and the failure attributed to the ground system, and (2) there was no way to determine the malfunctioning systems in the 472 firings for which the Douglas data were missing.

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III. NIKE OPERATION AND MAINTENANCE

The NIKE-AJAX is a ground-to-air missile approximately 33 feet long, including the booster section, and weighs about 2400 pounds. It has four components: missile guidance, hydraulic, propulsion, and warhead. The propulsion system is composed of a solid-propellant booster and a liquid-fuel rocket motor. The ground system is composed of a target-sequisition radar, a target-tracking radar, a missile-tracking radar, and a computer. The computer receives target-location information from the target-track radar and transmits instructions to the missile guidance unit via the missile-track radar.*

Since one of the two main variables studied here is the effect of missile usage on firing reliability, it is pertinent to describe briefly the NIKE's operational environment. When a missile is received on a tactical site, the battery, warheads, detonating cords, and fuel are installed to make it a "ready missile." There are normally about 30 ready missiles and one training missile on a tactical site. Missiles on a site are given daily, weekly, and monthly checkouts, and are also used in simulated combat exercises for crew training. The daily checkout is a visual check. The weekly one provides certain telerance checks which require the guidance and hydraulic units to operate for 5 to 10 minutes, while the monthly checkout operates the same systems 15 to 20 minutes. The combat exercises require elevation and handling of the missile, plus operation of the guidance and hydraulic units for about 5 minutes. On a training missile, these units may be operated as much as

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For a more complete exposition on the NIKE history, description, and operation see Bell Telephone Laboratories, Inc., Guided Missile System.

XSAM-A-7. Over-all System Operation, December 8, 1952, Revised May 8, 1954

(Confidential); Bell Telephone Laboratories, Inc., and Douglas Aircraft

Company, Inc., Project NIKE, History of Development, April 1, 1954 (Confidential).

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15 hours a month. In this study, usage is measured by the number of months on site and the number of hours the guidance unit is operated.

There are the usual three levels of maintenance — organizational, field, and depot. Organizational maintenance consists primarily of tolerance adjustments and the replacement of major components. A depot-rebuild was required after two years on a site during the period of this study; the current limit is 30 months. Also, the rebuilt missiles included in this study were seldom returned to a site prior to firing at Ft. Bliss. Currently, however, a substantial number of rebuilt missiles are placed on tactical sites prior to return to Ft. Bliss for firing.

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IV. MISSILE CONTINUITY

The missile does not necessarily keep the same components throughout its history. Guidance units, especially, are often replaced. Of those missiles which were fired without a rebuild, only 43 per cent still contained their original guidance units, and of those which were rebuilt only 23 per cent contained their original guidance units. However, there was 95-percent overall agreement between the rebuild status of the missile and of the guidance unit, as can be seen in the following table.

		Missile	i	
Missile Guidance Unit	Non-rebuilt	Rebuilt	Data Hissing	Total
Non-rebuilt Rebuilt Data Missing	1139 34 147	67 953 50	76 21 98	1282 1008 295
Total	1320	1070	195	2585

Similarly, the difference between the manufacturing dates for the missile and for its guidance unit was generally small: less than 150 days in 93 per cent of the cases. The rebuild dates were also within 150 days of each other in 93 per cent of the cases. For these reasons we have used single variables to define the age and rebuild status of the missile, rather than using one for the guidance unit and one for the missile. It should be pointed out, however, that since guidance units often become separated from missiles, the "time-on-site" measure does not necessarily apply to the guidance unit at the time of firing; and similarly, the measure of its usage does not necessarily apply to the rost of the missile.

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Excluding the guidance unit, in this discussion.

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V. METHOD OF ANALYSIS

The 2585 firings were divided into two approximately equal samples, called Samples I and II, on the basis of odd and even serial numbers. Sample I was used in an exploratory manner to form hypotheses on what variables or combination of variables were related to firing reliability. Sample II was not inspected until this step was complete, after which it was used to provide statistical tests of the hypotheses formed from Sample I. This procedure allowed us to examine many possible relationships in Sample I without impairing the validity of the statistical significance tests which were performed on selected relationships in Sample II.

The relations in both samples were examined through grouping the firing results according to both controlled variables and the variable of interest. The success ratios observed in the cells of the tables thus formed were converted into Chi-square statistics. These statistics are derived by comparing observed frequencies with those frequencies expected under the hypothesis that the variable of interest has not affected the data. Since the data in the tables are success ratios, the particular form of the Chi-square statistic known as the binomial index of dispersion is required.

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^{*}See, for example, W. J. Dixon and F. J. Massey, <u>Introduction to Statistical Analysis</u>, 2d ed., McGraw-Hill, New York, 1957, pp. 221-227.

^{**}P. G. Hoel, <u>Introduction to Mathematical Statistics</u>, lst ed., John Wiley and Sons, New York, 1947, pp. 196-197.

VI. RELATION OF DATE OF FIRING, DATE OF MANUFACTURE, FIRING CREW, AND REBUILD STATUS TO FIRING RELIABILITY

Figure 1 gives the proportion of successes in groups of 100 firings as a function of date of firing. Figure 2 provides the same presentation as a function of date of manufacture. Missiles manufactured after about December, 1954 (Missile Serial No. 4193) included a major modification, particularly in the guidance unit. The curves drawn in Figs. 1 and 2 were fitted to the data by an adjusted least-squares method described in Appendix A.

The preliminary analysis of the data (see p. 11) revealed a tendency for firing reliability to rise for later dates of firing; this tendency could not be accounted for by the intercorrelation of firing dates with other variables, such as dates of manufacture. In searching for an explanation, we found that around July, 1956, missiles began receiving a special painting which greatly reduced the propulsion failures caused by "burn-throughs" in the earlier firings. The following table provides the distribution of critical failures attributed to malfunctions in the propulsion system, as a function of date of manufacture and date of firing:

	Date of Firing											
	Apr 154 - Aug 156		Aug 156 - Oct 157		Oct #57 - Nov #58			Data Missing	Total			
Date of Manufacture		Propula. Failure	N	Propuls. Failure	N	Propuls. Pailure		Propuls. Failure	N	Propuls Failure		
Aug 152 - Apr 154 Apr 154 -	460	31	329	12	147	4	7	ı	943	48		
Dec 154 Dec 154 ~	297	9	362	13	275	4	9	1	943	27		
Peb 158	69	2	133	1	405	1	92	3	699	7		
Total	826	42	824	26	827	9	108	5	2585	82		

See Appendix B for a detailed presentation of the interrelations between date of firing and other variables.

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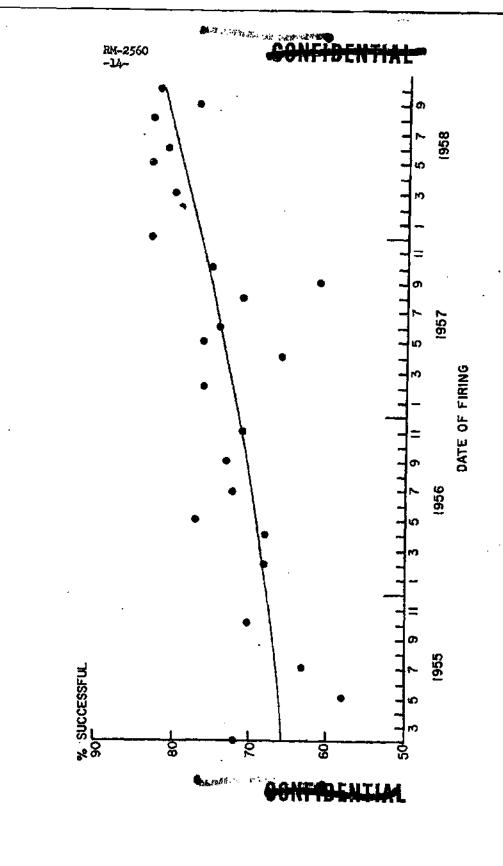


FIG. 1-RELIABILITY AS A FUNCTION OF DATE OF FIRING; DATA GROUPED IN SAMPLES OF 100 FIRINGS

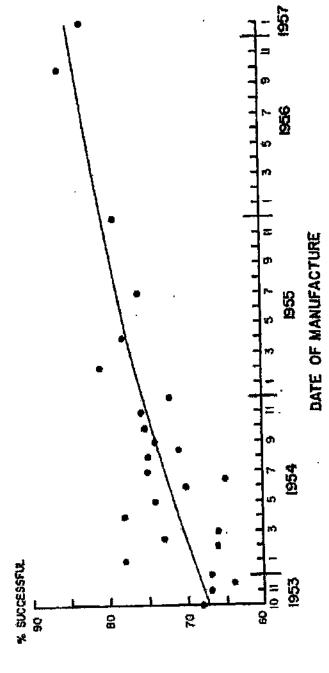


FIG. 2 - RELIABILITY AS A FUNCTION OF DATE OF MANUFACTURE: DATA GROUPED IN SAMPLES OF 100 FIRINGS

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Since the purpose of this analysis is to study the effects of age and usage, it is useful to control or eliminate secular trends which are due primarily to the date of firing. For this reason, the firings whose failures were attributed to propulsion malfunctions were eliminated from the sample, leaving a total of 2503 firings. Since we did not have information on system malfunctions for the 472 firings for which the Douglas data were not available, we did not eliminate propulsion malfunctions from this group. This creates some bias in the data, because as can be seen in Table 1, the firings for which the Douglas data are missing were primarily during the years 1957-1958. Thus, some missile failures due to propulsion malfunctions are probably included during this period. However, as can be seen from the above table on propulsion malfunctions, the rate of this kind of failure is so small for the later firings that the bias should be negligible. The remainder of this paper will be concerned with the sample of 2503 firings, i.e., those remaining after removal of the 82 missile failures attributable to known propulsion-system malfunctions.

Next, the effect of the type of crew conducting the firing was examined. Experienced Army crews from tactical sites fired approximately 58 per cent of the sample. These firings were called "annual service program rounds." Another 18 per cent were fired in training programs by inexperienced Army crews generally receiving their first experience in actual firings. Another 5 per cent were fired by contractor or Army Ordnance crews, and for 19 per cent the crew information was missing. The following table gives the success rate by crew type after removal of propulsion failures:

	<u>Firings</u>	% Successful
Annual service crew	1452	78.6
Training program	458	68.6
Contractor and Army Ordnance	121	76.0
Data missing	<u>472</u>	<u>74.8</u>
marrier o	2503	75+9

The training-program firings represented only a very small proportion of the total firings during the later firing dates, a fact which tended to raise the estimate of reliability as a function of firing date. In the remainder of this paper we shall treat the firing-crew variable in three groups: annual service, training program, and a combined group made up of the contractor, Army Ordnance, and firings for which the crew data were missing. Only the annual service group will usually be presented in tabular form, but when statistical significance tests are cited they are a result of all three groups in Sample II. The complete set of results is given in Appendix C.

Another variable closely related to date of firing is the remaild status of the missile. Only 9 per cent of missiles fired before May, 1957, were rebuilt, while about 81 per cent fired after this date were rebuilt before firing. Table 2 presents the relation between date of firing and success rate for missiles fired by annual service crews, with date of manufacture and rebuild status controlled. There is no longer a consistent trend in reliability as a function of date of firing, and the X² value for Sample II was not significant. Reading across the rows under the corresponding rebuild statuses of Table 2, one can still see evidence of increasing reliability as a function of later manufacture-dates for non-rebuilt missiles, while the reliability of rebuilt missiles rises only slightly. When date of firing

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Table 2

RELIABILITY FOR ANNUAL SERVICE CREWS AS A FUNCTION OF DATE OF FIRING;

DATE OF MANUFACTURE AND REBUILD STATUS CONTROLLED[®]

		Date of Manufacture												
	Aug	52 - J	pr 54		Apı	54 - 1	Dec 54	Dec 54 - Feb 58						
.	Non-Re	lon-Rebuilt Rebuilt			Non-F	lebuilt	Rebu	Non-R	ebuilt	Reb	1114			
Date of Firing	N	16 3	N	1 53	N	7 3	N	% 3	N	% 3	N	% 3		
Apr 54 - Aug 56	(162)b	72.3	(i)	°	212 (170)	80.2	1 (1)	_	31 (28)	90.3	(O)			
Aug 56 - Oct 57	64 (51)	79.7	119 (91)	76.5	91 (69)	75.8	91 (76)	83.5	75 (57)	76.0	10 (6)	_		
Oct 57 - Nov 58	(6)		109 (90)	82.6	(1)	-	125 (95)	76.0	131 (110)	84.0	102 (86)	84.		
Data Gesing	1 (0)		o (0)		1 (1)	-	(o)		1 (1)	_	(0)			
Total	289 (213)	73.7	231 (182)	78.8	305 (241)	79.0	217 (172)	79.3	238 (196)	82.4	112 (92)	82.		

^aFirings with missing data concerning rebuild status are not included in any of the tables throughout the remainder of the paper. As previously mentioned, failures attributed to propulsion malfunctions are excluded.

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The number in parenthesis indicates the number of successes.

OPercentages are not shown when N is less than 25.

is not controlled (total in Table 2), and all three groups of crews for Sample II are considered, missiles manufactured after December, 1954, were significantly more reliable (at the 95th percentile) than those manufactured prior to this date. The group manufactured between August, 1952, and April, 1954, did not prove significantly different from the April, 1954 - December, 1954 group in Sample II; the two are combined as a single group in the remainder of the analysis, and labeled the "old model." The "new model" comprises missiles manufactured after December, 1954 (after the major modification).

In Table 2, we may also examine the effect of rebuild status on reliability, with crew, date of manufacture, and date of firing controlled. The rebuilt missiles show a somewhat higher reliability here, and also for the other two crew types (not shown), but the difference did not prove to be significant in Sample II. It should be pointed out that the large majority of the rebuilt missiles in this sample experience only a storage environment between rebuild and firing, while the majority of the non-rebuilt missiles have some tactical-site environment between manufacture and firing.

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VII. RELATION OF THE MISSILE AGE FACTOR TO RELIABILITY

Table 3 presents reliability as a function of age at firing, with crew, model, and rebuild status controlled. The age referred to here is that from manufacture to firing, regardless of intervening rebuild. For non-rebuilt missiles, there is no difference in reliability as a function of missile age for the old model, and an apparent decrement for older missiles of the new model. However, the sample size for the latter is small, and the age-reliability relationship did not prove significant in Sample II. Similarly, no significant relationship was found between the age factor and reliability when both rebuilt and non-rebuilt statuses were examined. In view of the fact that the missiles were given a thorough field-level overhaul immediately before firing, the lack of statistical significance between the age factor and firing reliability is interpreted as follows: The effect, if any, that the age factor has on the missile reliability is negligible after a field-level overhaul.

Table 3

RELIABILITY FOR ANNUAL SERVICE CREWS AS A FUNCTION OF ACE OF MISSILE;

MODEL AND REBUILD STATUS CONTROLLED

			Old Mo	del		New Model						
	Non-Rebuilt			Rebuilt			Non	-Rebu	ilt	Rebuilt		
Missile Age (Days)	N	\$	%s	N	s	%S.	N	s	% S	N	\$_	% S
7 - 702	293	226	77.1.	2	2	_	182	156	85.7	21	18	_
702 - 1109	2 83	215	76.0	96	74	77.1	55	39	70.9	35	28	80. 0
1110 - 2002	16	12		350	278	79.4	0	0		56	46	82.1
Data Missing	2	1		0	0		l	1		0	0	
Total.	594	454	76.4	1,48	354	79.0	238	196	82.4	112	92	82.1



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It is interesting to examine the effects of the missile age factor on reliability, independent of the usage factor. We attempted to investigate the age-reliability relationship for non-rebuilt missiles which experienced only a storage environment between manufacture and firing, but the sample size and spread in age were not sufficient to measure any possible effect. In the rebuilt portion of the sample the effect of age may be examined independently of usage because only a small portion of these missiles experienced a site environment between rebuild and firing. The range in age from rebuild to firing is fairly small, however. Table 4 presents the reliability of rebuilt missiles as a function of date of rebuild, with crew and model controlled. Table 5 provides the reliability of rebuilt missiles as a function of age since rebuild at firing, with crew and model controlled. There was a slight trend for missiles with later rebuild dates to have a higher reliability, but this did not prove significant. There was no evident relationship between reliability and age of missile since rebuild. This was true even though there is a positive correlation between early rebuild date and age of missile since it was rebuilt (see Appendix B) which, in view of the slightly lower reliability for missiles with early rebuild dates, would tend to favor lower reliability as a function of missile age since rebuild. * Since the missiles received a thorough field-level maintenance inspection and repair as necessary immediately before firing, we may conclude that the effect, if any, between reliability and the missile age factor is negligible after a field-level overhaul.

White Sands Proving Ground disassembled, thoroughly inspected, and fired four missiles which had been in depot storage for from 27 to 46 months in order to examine the effect of storing missiles for long periods. Two of the four failed in flight, but it was concluded that the causes were not necessarily due to storage effects. Systems Test Div., White Sands Proving Ground, New Mexico, Shelf and Service Life of NIKE-AJAK Missiles, Test Plan 70, Technical Memo. 515, March, 1958 (Confidential).

Table 4

RELIABILITY FOR REBUILT MISSILES WITH ANNUAL SERVICE CREWS AS A FUNCTION OF DATE OF REBUILD; MODEL CONTROLLED

	01	d Mod	ol.	Ne	New Model			
Missile Rebuild Date	N	S	% S	N	S	* S		
Mar. 12, 1954 - Mar. 14, 1957	193	148	76.7	ខ	5	- · ·		
Mar. 14, 1957 - Sept. 20, 1957	189	153	81.0	15	13	_		
Sept. 20, 1957 - July 2, 1958	70	57	81.4	92	76	82.6		
Data Missing	25	18	1	28	21			
Total	477	376	78.6	143	115	80.4		

Table 5

RELIABILITY FOR REBUILT MISSILES WITH ANNUAL SERVICE CREWS AS A FUNCTION OF AGE SINCE REBUILD; MODEL CONTROLLED

Missile Age	OI	d Mod	e]	New Model			
Since Rebuild (Days)	N	\$	% S	rŧ	5	% S	
8 - 144 144 - 255 255 - 1467	128 175 145	102 136 116	79•7 77•7 80•0	64 37 11	56 26 10	87.5 70.3	
Data Missing	29	22		31	23	_	
Total	477	376	78.8	143	115	80.4	

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VIII. RELATION OF MISSILM USAGE TO RELIABILITY

Table 6 presents reliability as a function of number of months the missile was located on a tactical site, with crow, model, and rebuild status controlled. The data gave the time-on-site prior to rebuild, or since new if no rebuild occurred, but since only a small portion of the missiles in this sample were on a site subsequent to a rebuild, the variable includes essentially all the site-time experienced. If we assume that a rebuild eliminates any deterioration in reliability due to site environment, then we would expect to find the firing-success rate related to site-time in only the non-rebuilt portion of the sample. Table 6 shows a weak negative correlation between site-time and reliability. The X² value for Sample II was significant at the 86th percentile. As expected, there was no relationship between site-time and reliability for rebuilt missiles.

Table 6

RELIABILITY FOR ANNUAL SERVICE CREIS AS A FUNCTION OF TIME-CNTACTICAL-SITE: MODEL AND REBUILD STATUS CONTROLLED

			Old	Model	·	New Model							
Time Missile	Non-Rebuilt			13	Rebuilt			Non-Rebuilt			Rebuilt		
Was on Site (Months)	N	S	%s	N	S	xs.	11	s	೫೨	N	s	īs.	
0 - 0.8 0.8 - 13.0 13.0 - 48.0	147 248 81	122 186 62	83.0 75.0 76.5	52 130 198	1,2 102 160	80.8 78.5 80.8	150 28 49	129 21 39	86.0 75.0 79.6	14 13 71	11 11 57	- 20.3	
Data Missing	118	84		68	50		11	7		14	13		
Total	594	454	76.4	448	354	79.0	238	196	82.4	112	92	82.1	



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Table 7 presents reliability as a function of the amount of time recorded on the missile guidance unit since rebuild, or since new if no rebuild has occurred, with crew, model, and rebuild status controlled. There appears to be a negative correlation between reliability and the amount of time the guidance was used. The X^2 value for Sample II was significant at the 85th percentile. On examining Table 7, we see that there is little spread in time-on-the-guidance-unit for rebuilt missiles. If we obtain a X^2 value for only the non-rebuilt portion of the sample, the significance level rises to the 96th percentile.

Table 7

RELIABILITY FOR ANNUAL SERVICE CREWS AS A FUNCTION OF TIME MISSILE GUIDANCE
UNIT WAS OPERATED SUBSEQUENT TO REGUILD, OR SINCE NEW IF NO REBUILD

OCCURRED: MODEL AND GUIDANCE UNIT REBUILD STATUS CONTROLLED

Time Missile Ouidance Unit Was Operated (Hours)		Old Model						New Model					
	Non-Rebuilt ^a			Rebuilt			Non-Rebuilt			Rebuilt			
	N	\$	1 /S	N	ş	೫ಽ	11	ş	%s_	N	s	7.S	
0.4 - 3.2 3.2 - 7.0 7.0 - 257.9 ^b	69 155 297	54 123 220	78.3 79.4 74.1		204 91 16	79.1 78.4	91 97 77	77 84 55	84.6 86.6 71.4	65 30 6	52 26 4	80.0 86.7	
Data Missing	25	20		25	21		2	2	<u> </u>	2	2		
Total	546	417	76.4	421	332	78.9	267	218	81.6	103	84	81.6	

^{*}In Tables 7, 8, and 9 the rebuild status refers to the missile guidance unit. In previous tables it referred to the rebuild status of the remainder of the missile.

b92 per cent of this group is between 7.0 - 30 hours.

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Table 8 presents reliability as a function of total time the guidance unit was operated, both before and after rebuild for rebuilt missiles only, with crew and model controlled. No significant relationship is observed.

As mentioned earlier, only 43 per cent of non-rebuilt missiles and 23 per cent of those rebuilt contained their original guidance units at the time of firing. The time the guidance unit is operated, therefore, does not necessarily indicate the amount of usage the remainder of the missile has experienced in an appreciable portion of the sample. We attempted to examine the relationship between reliability and the amount of time recorded on the guidance unit for non-rebuilt missiles which retained their original guidance units. The results were of the same direction as for the total sample, but the sample size was too small to determine whether there was a stronger relationship under these conditions.

Table 8

RELIABILITY FOR MISSILES WITH REBUILT CUIDANCE UNITS AND ANNUAL SERVICE CREWS
AS A FUNCTION OF TOTAL TIME THE MISSILE GUIDANCE UNIT WAS OPERATED, BOTH
PRIOR AND SUBSEQUENT TO REBUILD; MODEL CONTROLLED

Total Time Missile Guidance Unit	01	d Mod	e <u>1</u>	New Model			
Was Operated (Hours)	N	S	% 3	Ħ	S,	% S	
0.7 - 6.6 6.6 - 14.3 14.3 - 245.5 ^a		51 125 97	72.9 80.1 79.5	12 33 51	9 25 44	 75.8 86.3	
Date Missing	73	59		7	6		
Total	421	332	78.9	103	84	81.6	

^{88%} of this group is between 14.3 - 40 hours.

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Table 9 presents the missile failures attributable to the guidance unit as a function of the amount of time recorded on the unit since rebuild, or since new if no rebuild has occurred, with rebuild status controlled. The data cover all three craw-types and both models. The percentages given in this table should not necessarily be interpreted as direct reflections of guidance-unit reliability, since the system causing the failure was listed as unknown in approximately 45 per cent of the cases. The trend is for the number of failures attributed to guidance-unit malfunctions in non-rebuilt missiles to be positively correlated with the amount of time the unit was operated. The relationship did not, however, test significant in Sample II. There is no relationship evident between the number of guidance-unit malfunctions and operating time since rebuild. It should be observed that this table is not strictly comparable with previous results, since crew type and model have not been controlled. The total time registered on the guidance unit on rebuilt missiles (not shown) showed no relationship to the number of guidance-unit malfunctions.

It is interesting to note in Table 9 that while the overall reliability of rebuilt missiles is generally as high or higher than that of non-rebuilt missiles, the reliability of the guidance unit appears to be appreciably lower in the rebuilt missiles. This difference is significant at the 96th percentile in Sample II. As mentioned before, this apparent decrement in the guidance-unit performance does not show evidence of being related to the time the unit was operated prior to rebuild.



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Table 9

THE RELATION OF KNOWN MISSILE CUIDANCE UNIT PAILURES TO THE AMOUNT OF OPERATING TIME ON THE GUIDANCE UNIT SUBSEQUENT TO REBUILD OR SINCE MEN IF NO REBUILD OCCURRED: GUIDANCE UNIT REBUILD STATUS CONTROLLED.

Time Guidance Unit Was	Non-	Rebui	1t ·	Rebuilt			
Operated (Hours)	Nb	8	% S	Ŋb.	S	%S	
0.4 - 3.2 3.2 - 7.0 7.0 - 257.9°	146 269 428	141 256 402	96.6 95.2 93.9	295 167 38	269 152 36	91.2 91.0 94.7	
Data Missing	66	61		20	29		
Total	909	860	94.6	530	486	91.7	

Includes all three types of crews. Does not include firings for which rebuild-status data were missing.

bExcludes firings which failed because of non-guidance-system malfunctions; also excludes firings for which the Douglas data (system-malfunction information) were net available.

*86% of this group is between 7.0-30 hours.

Finally, an effort was made to examine, over the total sample, the reliability of non-rebuilt missiles which had more than 30 hours of operating time on the guidance unit. The 53 missiles in this category had a success rate of 67.9 per cent, as compared to 75.0 per cent for the 1182 non-rebuilt missiles with less than 30 hours of guidance operation. While this result is in the expected direction, the difference was not significant for this size of sample. Rebuilt missiles which had more than 40 hours of total guidance operating time were similarly examined. The 46 missiles in this category had a success rate of 76.1 per cent, as compared to 77.9 per cent for the 928 rebuilt missiles with less than 40 hours total guidance operating time. The difference is clearly insignificant.

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In summary, the results of this study appear to support the hypothesia that even after a field-level overhaul, prior missile usage, as measured here, is generally negatively correlated with firing reliability. The number of months the non-rebuilt missile was on a tactical site showed a weak negative correlation, and the number of operational hours on non-rebuilt missiles evidenced a somewhat stronger negative relationship. Because of the divorce rate between the guidance unit and the remainder of the missile, the time-onsite measure is strictly applicable to only 43 per cent of the non-rebuilt guidance units. On the other hand, while the operational time on the guidance unit probably gives a good measure of the usage of the guidance unit, it is indicative of usage of the remainder of the non-rebuilt missile in only 43 per cent of the cases. If we eliminate both ground- and propulsion-system failures, the guidance unit accounts for approximately oil per cent of the remaining known causes of failures. This fact, together with the relationship to reliability found in this paper, indicates that the time the guidance unit is operated may be a better measure of usage than the time-on-site.

An alternative hypothesis to explain the negative correlation between guidance-unit operating time and reliability is that inherently inferior missiles require more time for the guidance unit to be calibrated, and this inferior quality is still present when the missiles are fired. We attempted to check this possibility at a NIKE site. One of the findings was that the frequency with which tolerance adjustments were required at checkout increased as a function of the number of checkouts undergone. This tends to support the conclusion that the guidance-unit operating-time is a valid measure of missile usage.

IX. RELIABILITY AS A FUNCTION OF CHECKOUT FREQUENCY

As mentioned in the Introduction, we do not have snough information from the historical summaries to determine optimal missile checkout frequencies in this study. However, it may prove useful to present a simplified model of how one would use such information to develop preferred maintenance policies. When a checkout is performed on a tactical missile, it is implicitly assumed that there is some chance of a failure occurring during the standing period since last checkout. If we make the usual assumption that the time-to-failure is defined by the negative exponential distribution, the probability of surviving t hours of standing time may be expressed by $e^{-t/\frac{T}{t}}$, where t represents the mean-standing-time-to-failure. In this study, we could not assess this "temporary" effect of standing time on tactical missiles. The effect of standing time on checkout reliability has been experimentally measured on other missiles, however.

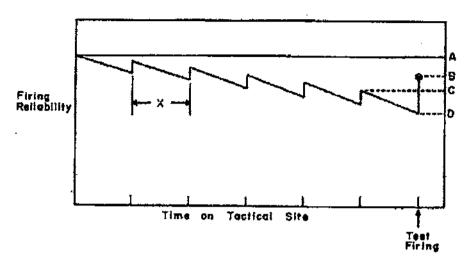
This study found that use of the missile, as defined by operation of the guidance unit for checkout and other purposes, was negatively correlated with firing reliability even after the performance of a field-level overhaul. When we introduce this factor, along with an assumption about the probability of surviving the standing time between checkouts, we may draw the hypothetical diagram seen on the next page.



^{*}See, for example, Final Report on the Employment and Suitability Test of the GAR-1 (Falcon) Missile, Hq. Air Proving Grounds, Eglin AFB, Florida, January, 1957, p. 15; and A. L. Story, "Military Implications of Guided Weapons Reliability" (U), Proceedings of the Joint Military-Industry Guided Missile Reliability Symposium, Redstone Arsenal, Huntsville, Alabama, Vol. 1, October, 1956 (Secret).

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The missile is assumed to have reliability A when delivered to the site. The time between checkouts is X and reliability is assumed to fall by an amount, C-D, during X hours of standing in a ready state. During the time from delivery to the test firing, the reliability drop due to usage is A-C. We will assume that the effect of age independent of usage is negligible here. The reliability increase represented by B-C is intended to represent the effect of the field-level maintenance the missiles in this sample receive prior to firing. The value A-B represents the usage effect remaining after the field maintenance, which was the quantity measured on the NIKE sample.

If we adopt this very simple model of reliability of the tactical missile, and assume that the missile remains on a site for a fixed period without receiving a depot-rebuild, we may solve for an optimal checkout frequency. Such a policy would simply maximize the area under the reliability function.



In actuality, the factors affecting missile combat reliability are much more complicated than as outlined here. The reduction in firing reliability as a function of usage may not be linear, and three other factors which are important are not considered: the cost of maintenance, missile time off alert due to checkcuts and resulting maintenance, and use of the missile for training.



^{*}McGlothlin and Yorshis, op. cit. (Cf. footnote, p. 1.)

^{***}For a model which considers some of these factors, see Eloise E. Bean and W. H. NcGlothlin, A Model for Assessing the Effect of Maintenance on Missile Reliability, The RAND Corporation, Research Memorandum RM-2451, September 23, 1959. See also R. S. LaValles and D. S. Stoller, The Effect of Maintenance and Reliability on the Operational Effectiveness of an Interceptor Squadron: A Case Study (Project LOCK-ON) (II), The RAND Corporation, Research Memorandum RM-1499, June 8, 1955 (Secret - Limited Distribution).

X. CONCLUSIONS

The primary conclusions of this study are that within the range of age and usago factors studied here, (1) there is evidence that, even after a field-level overhaul, prior missile usage on tactical sites, particularly as measured by the guidance-unit operating-time, reduces firing reliability; (2) the effect, if any, of missile age factors on reliabilit is negligible after a field-level overhaul, and particularly, there appears to be no postoverhaul effect of prior time in depot storage; and (3) with the exception of missile-guidance malfunctions, the reliability of rebuilt missiles compares quite favorably with those not rebuilt.

These results should be interpreted in the light of maintenance policies which provide a field-level maintenance inspection and repair-as-necessary on the NIKE immediately prior to firing. It is probable that the unfavorable effects of missile age and usage on combat reliability are greater than those measured here; but the intervening field-level overhaul between removal from site and firing allows us to measure only the residual effects subsequent to this maintenance action. Test firings are normally conducted only after very thorough checkout and maintenance procedures. Tactical missiles, of course, would not have the benefit of this sort of maintenance immediately before firing.

This points up a problem that is common to most missile programs. The use of test firing results can lead to unrealistic estimates of combat capatility, and can tend to obscure empirical measures of the effects of age and usage on firing reliability which, in turn, would be useful in determining preferred maintenance procedures for combat missiles. At the time a missile weapon system is considered operational, it would seem advisable to initiate

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test firings under conditions (1) which more closely simulate those of a combat environment, and (2) such that the effects of pre-launch environments on missile reliability can be more accurately assessed.*

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See D. S. Stoller, The Measurement of Missils Reliability in Pre-Launch Operating Environments, The RAND Corporation, Research Memorandum RM-2508, January 1, 1960.

Appendix A

METHOD OF DETERMINING RELIABILITY TREND

In Figs. 1 and 2, the firing success rates are plotted by groups of 100 firings against the date of firing and the date of manufacture, respectively. It is evident by inspection that a secular trend exists in the data. A function of the form

 $Y = A + BX + CX^2$.

Where

Y = success rate

X - date

A, B, C = fitted constants,

was considered adequate to represent the trend within the range of the data for each figure. Since the success rates observed in the data are not random observations from homogeneous binomial populations, the least-squares method for calculating fitted constants was adjusted to account for the heterogeneity. The adjustment was made by (1) estimating the standard deviation of each group, (2) weighting each observed-success rate by dividing by the estimated standard deviation, (3) obtaining a least-squares fit on the weighted success rates, and using step (3) to form a new estimate of the standard deviation for each group and reiterating a suitable number of times. The calculating procedure is given below:

Let: Ic(X) = observed success rate for date group X;

 $T_{C}^{(1)}(x)$ = calculated success rate for date group X, i-th iteration;

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$$A^{(1)}$$
, $B^{(1)}$, $C^{(1)}$ = least-squares coefficients, 1-th iteration; and $Z^{(1)}_{o}$ (X) = weighted observed success rate for date group X, i-th iteration.

(1) Calculate least-squares coefficients A_0 , B_0 , C_0 with $Y_0(X)$ and Y_0

(2)
$$Y_0^{(1)}(x) = A_0 + B_0 X + C_0 X^2$$

(3) $Z_0^{(1)}(x) = Y_0(x) / \sqrt{Y_0^{(1)}(x) \left[1 - Y_0^{(1)}(x)\right]}$

(Since the groups are equal in size, the factor for the number of observations in the group may be dropped without affecting the results.)

(4) Calculate least-squares coefficients A_1 , B_1 , C_1 with $Z_0^{(1)}(X)$ and X.

(5)
$$2c^{(1)}(x) = a_1 + b_1x + c_1x^2$$

(6)
$$Y_C^{(2)}(x) = Z_C^{(1)}(x)$$
 $\sqrt{Y_C^{(1)}(x) \left[1 - Y_C^{(1)}(x)\right]}$

(7)
$$z_o^{(2)}(x) - Y_o(x) / \sqrt{Y_c^{(2)}(x) \left[1 - Y_c^{(2)}(x)\right]}$$

(8) Calculate least-squares coefficients A_2 , B_2 , C_2 with $Z_0^{(2)}(X)$ and X.

(9)
$$2c^{(2)}(x) = A_2 + B_2x + C_2x^2$$
,

etc.

The iteration is terminated when

$$\begin{array}{c|c} \max & Y_{C}^{(i)}(x) - Y_{C}^{(i-1)}(x) & \leq \text{desired accuracy.} \end{array}$$

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Appendix B

INTERRELATIONS OF ACR AND USAGE VARIABLES

Table 10 provides the interrelations of the variables studied in this paper. Propulsion malfunctions are included. The following is the code to the column headings:

- A. Number of missiles in group
- B. Percentage of successful firings
- C. Percentage manufactured from August, 1952 to August, 1954
- D. Percentage fired from April 27, 1954 to August 5, 1957
- E. Average age of missile at firing in days (firing date minus manufacturing date)
- F. Percentage of missiles that were rebuilt
- G. Percentage of missiles that were rebuilt from December 3, 1954 to June 7, 1957
- H. Average age of missile subsequent to rebuild in days (firing date minus rebuild date)
- I. Time missile was on tactical site in months
- J. Total time missile guidance unit was operated in hours since manufacture
- K. Time missile guidance unit was operated since rebuild, or since manufacture if no rebuild occurred
- L. Percentage of missiles fired by annual service crews
- M. Percentage of missiles fired by Army Ordnance craws
- N. Percentage of missiles fired by field training program craws
- O. Percentage of missiles fired by contractor crews
- P. Percentage of missiles for which crew data were missing

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For footnotes, see p. 42.

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4 5	0 - 19.0	%%	Ċ.	J. F.		1301	ķ	85	276	15.8	15.	10 v	80.	88	ដូ	8	ų.
}			:]	1				•	~-	:			•	?	•	?	•
Dat	Data Mesing	208	69•	36.	36	834	•39	.28	797	1	13.6	80	747	8	ন	ಕ	88

Table 10 -- continued

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7				4 8	1_	85.		<u> </u>		85.	88888 88888	-
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-	1			45	1	13.8	ľ	10.11. 10.11.		13.8	21318 1.2011	
H	1			7. 2.5.	1 -	9.6		20 H 0		8 9.6	99 8 8 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	-
0				22.20	T	.22 228		1967 1967		.22 228	25.25 25.25 25.25 25.25 24.25	
54			_	£ 69	\vdash	54.5		1843	_	.45	<u> </u>	L
Ω	. <u>8</u>	28	1020	501	877	763	985	798 783 878	728	691	670 726 801 237 1105	L
0		_		.56 .47		.55 50			+ 1	.55 .50	.58 .52 .83 .88 .07 .92 .26 .08	1
В			_	73		.74		£43		.74	5. 25. 25. 25. 25. 25. 25. 25. 25. 25. 2	-
4	337	Ŕ	333	337	563	2565	38.88	367	383	2585	1493 494 117 472	2000
^	Total Guidance Unit, Operating Time (Hrs) ⁵ .7 4.3 6.6	ı	1 4.7	. [Data Mesing	Total	19 g.	4.5 - 13.2 - 25	Data Missing	Total	Grew Type Annual Service Army Orchance Fleid Training Prog. Contractor Crew Data Missing	
	Pota Open	•	***	크제	# a-] 		o EM		1	9 4 4 E 3 5	

See p. 39 for key to boxheads.

**Dotal time since name acture.

**So per cent of this group is between 21 and 40 hours.

**Alme since rebuild or since name acture if no rebuild occurred.

**Sk per cent of this group is between 13 and 30 hours.



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Appendix C

SUPPLEMENTARY TABLES

Tables 11-17 correspond to Tables 2-8 in the text and provide complete data for the three crew groupings: Annual Service, Training Program, and others. The last group comprises Contractor and Army Ordnance crews, and novera those firings for which crew data were missing. These tables include the data for firings with unknown rebuild status, which were emitted in Table 2-8. Missile firings which failed because of malfunctions in the propulsion system have been deleted from these tables.

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Teble 11

RELIABILITY AS A FUNCTION OF DATE OF FIRTHS: CREM, DATE OF MANUFACTURE, AND REBUILD STATES COFFICIED

			Aug	Aug 52-Apr	武			₽	35.	St. Dec Sk				3	Dec 54-Feb 58	98		
		Hog-repuilt		Rebut 14		Rebutld Status Missing		Bon-rebuilt		Rebut 14	Bebuild Status Missing	1 3 4	Hou-Rebuilt	Part 14	Rebutlt	ដ	Rebuild Status Maning	E E E
Crew Type	Date of Piring	末	3	я	6	E 11	E.	8	Α.		-	22	=	92	=	ď	-	8
Annual Service	Apr 54-Aug 56 Aug 56-Oct 57 Oct 57-Boy 58 Data Masing	₹ 3 0°	397 0 0	, H. C.	448°	and on the	212 42	5844	٠ 4 <i>4</i> 000	45%0	ogon	0000	ಇಕಷ್ಟ	1 1 2 2 B	ဝဠဋ္ဌာ	ഠയയ്ക	0 12 2 ±	odus
	Total	88 88	213	231	183	15 1	305	142	717	27.		្ន	238	367	द्या	8	ಸ	23
Training Program	Apr 54-Aug 56 Aug 56-Oct 57 Oct 57-How 58 Data Massing	25. 2. 2. 3.	8500	ಇಕ್ಷಗಾಧ	0540	mw00	m+00	9800	0 × ~ 0	9500	0.400	0 70 0	5000	స్తం	0000	0000	0400	0#00
	Total	913	137	8	3	æ	7 207	33	3	ಸ	*	3	17	1.5	0	0	~	*
Contractor, Army Ordnance, and Grew Data Massing	Apr 54-Aug 56 Aug 56-Oct 57 Oct 57-Nov 56 Data Missing	สูดฯ๐	Ä-100	୦ ମୁଖ୍ଜ ୦	೦೫೪೦	moon	40,0 m	+V-10 410	0.65	രക്ളം	ספיני	9846	^ទ ង់ខ្លួ	ma X o	० ध्रुटा ०	ဝက္ရမ	Br_u\$	ผพพส
	Totel	\$3	Ħ	7	×	4	8	9		390	6	7	1.7	×	335	9	ğ	8

Table 12

RELIABILITY AS A FUNCTION OF ACE OF HISSILE; CREM, MODEL, AND REBUILD STATUS CONTROLLED

	Rebuíl d Status Hissing	S	800 K	23	0400	79	\$00g	23
	Rebuil Status Missin	175		R	0 10 0	7	<u> </u>	1307
н	ii t	177	#830	8	0000	٥	456	102
Her Model	Retuil t	. 23	식상였다	7	0000	٥	4540	135
He	built	га	39	196	स०००	25	¥400	35
	lon-Rehuilt	И	182 55 0	238	17 0 0	17	400	7.7
	1.d 1.d 1.g	S	4v0v	22	4400	12	9999	п
<u> </u> 	Rebuild Status Missing	15	~~~~	29	6 r 40	2	4000	7
	指	S	77 278 0	354	0220	8	25.50	182
Old Model	Rebuilt	æ	2 96 350 0	844	1 33 67 0	ig ig	0寸20~	24.5
큥	milt	82	225 215 12	454	304 404 6	8	5,640	23
	Non-Rebuilt	Æ	293 283 16 2	594	0 TI 591 271	323	ద్దాం	4.5
		Missile Age (Days)	7- 702 702-1109 1110-2002 Data iissing	Total	7~ 702 702-1109 1102-2002 Data Assing	Total	7- 702 702-1109 1110-2002 Data Missing	Total
		Crew Type	Annal Service		Training Program		Contractor, Arry Ordnance, and Grew Data Pissing	-

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Table 13

RELIABILITY FOR REBUILT MISSILES AS A FUNCTION OF DATE OF REBUILD: CREW AND MODEL CONTROLLED

		Old 1	lodel.	Nev	/ Model
Crew Type	Missile Rebuild Date	N	s	N	8
Annual Service	Mar 12, 1954 - Mar 14, 1957 Mar 14, 1957 - Sep 20, 1957 Sep 20, 1957 - July 2, 1958 Data Missing	193 189 70 25	148 153 57 18	15 92	5 13 76 21
Training Program	Mar 12, 1954 - Mar 14, 1957 Mar 14, 1957 - Sep 20, 1957 Sep 20, 1957 - July 2, 1958 Data Missing	83 18 0 12	65 15 0 10	 	0 0 4
and Crew Data	Mar 12, 1954 - Mar 14, 1957 Mar 14, 1957 - Sep 20, 1957 Sep 20, 1957 - July 2, 1958 Data Missing	47 123 75 14 259	33 85 64 11	13 5 118 106	11 2 95 81

COMPREHENTIAL

Table 14

RELIABILITY FOR REBUILT MISSILES AS A FUNCTION OF ACE SINCE REBUILD: CREW AND MODEL CONTROLLED

		O1 Mod		Ne Hod	
Стем Туре	Missile Age Since Rebuild (Days)	H	8	Ŋ	3
Annual Service	8 - 144 144 - 255 255 - 1467 Data Nissing	128 175 145 29	102 136 116 22		56 26 10 23
	Total	477	376	143	115
Training Program	8 - 144 144 - 255 255 - 1467 Data Missing	31 24 46 12	23 19 38 10	0 0 5	0004
· · · · · · · · · · · · · · · · · · ·	Total	113	90	5	4
Contractor, Army Ordnance, and Grow Data Missing	8 - 144 144 - 255 255 - 1467 Data Missing	42 77 126 14	27 61 94 11	82 32 21 107	63 26 18 82
	Total	259	193	242	189

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UNCLASSIFIED/LIMITED

HELIABILITY AS A FUNCTION OF THE ON TACTICAL SITE; CREM, MOHEL, AND RESULD STATES CONTROLLED

Table 15

			8	Old Model	7				×	New Model	- H		
		Non-rebuilt	built	Rebuilt	114	Rebuild Status Missing	12 a 2.	Non-rebuilt	Suilt	Rebuilt	#	Rebuild Status Missing	D d d d d d d d d d d d d d d d d d d d
Crew Type	Time Missile Was On Site (Months)	N	S	N	Ø	=	'n	121	r)	z	တ	×	S
Annal Service	0 - 0.8	87Z	23%	22,23	702	0.8	0.00	ន្តន	84	ដង	គគ	£12	25,
	Data Missing	311	70.62	86	इस	2	~¤	3 7	8,5	다크	K D	~~ -4	~-
	Total	765	757	844	354	29	22	238	196	या	g	류	ຄ
Training Program	0 - 0.8 0.8 - 13.0 13.0 - 46.0 Data Missing	3885	44234	នេះក្នុង	ដ្ឋមន្ត្រ	7 7 7	ччию́	16 0 0	컴퓨၀၀	0000	0000	04H0	0400
	Total	333	8	101	.8	22	OI	17	15	٥	0	5	-4
Contractor, Amy Ordnance, and Grew Data Hissing	0 ~ 0.8 0.8 - 13.0 13.0 - 48.0 Data Missing	24~4	Bau.	ಬ್ಯಾಚ್ವಿಗ	ងឧឌឌ	หออหู	800B	35	#uow	35 g	048n	18200	почё
	Total	4.5	12	24.5	182	7	п	47	Ж	135	107	701	器

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RELLABILITY AS A FUNCTION OF TIME MISSIES CUIDANCE UNIT WAS CETATED SUBSEQUENT TO REBUILD, OR SINCE NEW IF NO REBUILD OCCURRED; CREM, HOME, AND CUIDANCE-UNIT REBUILD STATIS CONTROLLED

Table 16

			8	Old Model	굨				ar.	New Model	77		
,	The Guldance	Non-r	Ron-rebuilt	Rebu	Rebuilt	Rebuil Status Masin	Rebuild Status Masing	llon-re	Hon-rebuilt	Reby	Rebuilt	Rebuild Status Missing	50 Sign
Grew Type	Unit Has Operated (Hours)	11	Ω.	N	S	×	Ø	ø	w	=	s	in in	s
Annual Service	0.4 - 3.2 3.2 - 7.0 7.0 - 257.98 Data idssing	69 155 297 25	≠ង្កង្គង	23 116 22 23	ಕ್ಷಜನಚ	48.8 48.8	40.4 ₹	22.52	E8%°	స్ట్రాంగ	₩\$.44	иннь	ныча
	Total	34.6	417	421	332	701	81	267	23	203	ਡੋਂ	ដ	۵۰
raining Program	0.4 - 3.2 3.2 - 7.0 7.0 - 257.9ª Data Edseing	######################################	~ខ្លះដ	ឧ≇ส∽	25,52	9½ 17,	0448	بر و و	Hudo	0000	0000	оннн	Оннн
	Total	263	173	8	69	ដូ	53	19	77	٥	0	3	e,
Contractor, Army Ordnance, and Gres Data Missing	0.4 - 3.2 3.2 - 7.0 7.0 - 257.9ª Data Missing	4절정니	여 자 차 명 다	105	\$89, 11,4	40년	% °°	7 77 33	পরণ্য	844°	25 A	ча аБ	मलल द्ध
	Total	53	ನ	23,	173	17.	13	87	%	72	100	78	53

Approximately '92 per cent of these groups are between 7 and 30 hours.

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RELIABILITY FOR MISSILES WITH REBUILT GUIDANCE UNITS, AS A FUNCTION OF TOTAL TIME GUIDANCE UNIT WAS OPERATED BOTH BEFORE AND AFTER REBUILD; CREA AND MODEL CONTROLLED

Table 17

			l b	Old :lodel	To To		Г			New Model	le!		
٠	Total Time Guidance	:lon~rebuilt	built	Reh	Rebuilt.	Rebuild Status Missing	ild ing	Won-rebuilt	built	Rebuilt	ut.t	Rebuild Status Missing	11d
Crew Type	Operated (Hours)	Ħ	. თ	k=	ഗ	‡#.	က	缸	5/2	×	Ø	=	S
Armual Service	0.7 - 6.6 6.6 - 14.3 14.3 - 245.5	445	165 118 103	533	42,8	440	13.7	174 58 16	12.1 43.0	ಚರ್ಜ	25.	ÒНг	0 4-
	Data Massing Total	39	ह्य <i>भ</i>	23	33.2	& 5	68	19	`큐 ;	(r ;	, ۷	10 ;	46- (
Training Program	0.7 - 6.6 6.6 - 14.3 14.3 - 245.5a	23 23 28	386	32°	~28	OHO	OH	"P	2 280	9 000	00	1 70	~ ~ o
	Data Mesing Total	71. 263	% %	≀র ৪	ଜ ନୀ	18 A	8 8	24 61	, y	000	00 0	0 % ~	ON #
Contractor, Arry Ordnance, and Crew Bata	0.7 - 6.6 6.6 - 14.3 14.3 - 245.53 Data Musaing	837.	11 7	3228	2223	OHOŽ	040	220	ጸውጣ	428,	57.2	нон	нон
	Total	જ	*	2	173	17	1 12	\$	₹ \$	• हे	^ 8	% %	% A

Approximately 88 per cent of these groups are between 14 and 40 hours.

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